ISTANBUL TECHNICAL UNIVERSITY GRADUATE SCHOOL OF SCIENCE ENGINEERING AND TECHNOLOGY



MKC525E Finite Element Analysis in Engineering

Homework 4

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Figure 1: Geometry used in Question 1.



Figure 2: Boundary conditions in Question 1.



(a) Coarse mesh

Figure 3: Example for the different mesh sizes used in Question 1. 2D, Plane stress, linear quadrilateral (PLANE182).







Figure 5: von-Mises stress for the mesh sizes given in figure above.



Figure 6: Number of elements as a function of element size.



Figure 7: Deformation as a function of number of elements.



Figure 8: von-Mises stress as a function of number of elements.

Linear beam, shell and solid elements are compared in this question. Figures for deformation and stress are given in Appendix.



Figure 9: Boundary conditions in Question 2.



Figure 10: Example for types of mesh used in Question 2 (biggest mesh size).







Figure 12: Deformation as a function of number of elements (shell and solid elements).



Figure 13: Deformation as a function of number of elements (beam elements).



Figure 14: von-Mises stress as a function of number of elements (shell and solid elements).



Figure 15: Deformation for the mesh sizes given in figure above (biggest mesh size).



Figure 16: Deformation for the mesh sizes given in figure above (smallest mesh size).



Figure 17: von-Mises stress for the mesh sizes given in figure above (biggest mesh size).



Figure 18: von-Mises stress for the mesh sizes given in figure above (smallest mesh size).



Figure 19: Closer look at von-Mises stress for the smallest mesh sizes given in figure above.

There is an ideal geometry in this question. So there is no distorted elements. Temperatures and heat flux in x-direction is same for every element size (200 mm(1 element), 2 mm(130,000 elements)). Heat flux in y and z-directions are changing with the element size but they are considered numerical errors $(< 10^{-7})$.



Figure 20: Boundary conditions in Question 3.



Figure 21: Two different mesh for the problem.



Figure 22: Temperatures.



Figure 23: Heat flux (x-direction).



Figure 24: Boundary conditions in Question 4.



Figure 25: Deformation for first three eigenvalues (Load Multipliers) (Quadratic hexahedral).



Figure 26: Number of elements as a function of element size.



Figure 27: First eigenvalues as a function of number of elements.



Figure 28: Second eigenvalues as a function of number of elements.



Figure 29: Third eigenvalues as a function of number of elements.

In Ansys Workbench Mechanical, 2D surface geometry can not be divided into layers. So in this section. 3D linear tetrahedral, hexahedral and quadratic hexahedral elements are used to compere two and four layers. Comperations made in this section:

- 2 layers vs. 4 layers for quadratic hexahedral (first 5 constrained mode frequencies),
- Linear hexahedral vs. quadratic hexahedral for 4 layers (4. and 5. constrained mode frequencies)
- Linear tetrahedral vs. linear hexahedral for 4 layers (1., 2., 4. and 5. constrained mode frequencies)



Figure 30: Geometry used in Question 5.



Figure 31: Example for the two and four layers used in Question 5 (5 mm element size) (Linear hexahedral).



Figure 32: Example for the two and four layers used in Question 5 (5 mm element size) (Linear tetrahedral).



Figure 33: First 12 natural frequencies for normal (blue) and constrained (orange) modes.



Figure 34: 2 layers (blue) vs. 4 layers (orange) for quadratic hexahedral.



Figure 35: 2 layers (blue) vs. 4 layers (orange) for quadratic hexahedral.



Figure 36: Linear hexahedral vs. quadratic hexahedral for 4 layers.



(a) 1. mode

(b) 2. mode

Figure 37: Linear tetrahedral vs. linear hexahedral for 4 layers.



Figure 38: Linear tetrahedral vs. linear hexahedral for 4 layers.

5.1 Deformations



Figure 39: Deformations are given with this mesh (5 mm Linear tetrahedral).



Figure 40: Deformation for 1. modes (Linear tetrahedral - four layer).



Figure 41: Deformation for 2. modes (Linear tetrahedral - four layer).



Figure 42: Deformation for 3. modes (Linear tetrahedral - four layer).



Figure 43: Deformation for 4. modes (Linear tetrahedral - four layer).



Figure 44: Deformation for 5. modes (Linear tetrahedral - four layer).



Figure 45: Deformation for 6. modes (Linear tetrahedral - four layer).



Figure 46: Deformation for 7. modes (Linear tetrahedral - four layer).



Figure 47: Deformation for 8. modes (Linear tetrahedral - four layer).



Figure 48: Deformation for 9. modes (Linear tetrahedral - four layer).



Figure 49: Deformation for 10. modes (Linear tetrahedral - four layer).