## Istanbul Technical University

Graduate School of Science Engineering and Technology



## MKC517E Special Topics in Solid Mechanics

## Final

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1

A flat plate with a through-thickness crack (Fig. 1.8) is subject to a 100 MPa (14.5 ksi) tensile stress and A flat plate with a through-thickness crack (Fig. 1.8) is subject to a 100 MPa (14.5 ksi) tensile stress and<br>has a fracture toughness  $(K_{IC})$  of 50 MPa $\sqrt{m}$  (45.5 ksi $\sqrt{m}$ ). Determine the critical crack length for this assuming the material is linear elastic.



Figure 1: Finite width plate with a through crack at the center of plate.

$$
K_{\rm IC} = K_{\rm I} = \sigma \sqrt{\pi a_{\rm c}} \tag{1.1}
$$

$$
50 = 100\sqrt{\pi a_C} \tag{1.2}
$$

$$
a_c = 79.58 \text{ mm} \tag{1.3}
$$

So total crack length  $2a_c$  equals to  $159.15\,\mathrm{mm}.$ 

Compute the critical energy release rate  $\mathcal{G}_c$  of the material for E = 207000 MPa

$$
\mathcal{G} = -\frac{d\Pi}{d\mathbf{A}} = \frac{\pi\sigma^2 a}{E} \tag{1.4}
$$

$$
\mathcal{G}_{\rm c} = \frac{\rm K_{IC}}{\rm E} \tag{1.5}
$$

So,

$$
\frac{50^2}{207000} = 0.012077 \text{ MPa} \cdot \text{m} = 0.012077 \text{ J} \cdot \text{mm}^{-2}
$$
 (1.6)

$$
\mathcal{G}_{\rm c} = 12.077 \,\mathrm{kJ \cdot m^{-2}}\tag{1.7}
$$

A material exhibits the following crack growth resistance behavior:

$$
R = 6.95 (a - a_0)^{0.5}
$$
 (2.1)

where  $a_0$  is the initial crack size. R has units of kJ · m<sup>-2</sup> and crack size in millimeters. Elastic modulus of this material  $E = 207000 \text{ MPa}$ . Consider a wide plate with a through crack  $(a \ll W)$  that is made from this material.

If this plate fractures at 138 MPa, compute the following:

The half crack size at failure  $(a_c)$ 

The conditions for stable crack growth can be expressed as follows:

$$
\mathcal{G} = R \tag{2.2}
$$

and

$$
\frac{d\mathcal{G}}{da} \le \frac{dR}{da} \tag{2.3}
$$

Unstable crack growth occurs when

$$
\frac{d\mathcal{G}}{da} > \frac{dR}{da} \tag{2.4}
$$

$$
\mathcal{G} = \frac{\pi \sigma^2 a_c}{E} = 6.95 (a_c - a_0)^{0.5}
$$
\n(2.5)

$$
\frac{dG}{da} = \frac{dR}{da} \tag{2.6}
$$

$$
\frac{\pi \sigma^2}{E} = 3.457 (a_c - a_0)^{-0.5}
$$
\n(2.7)

 $\sigma = 138 \text{ MPa}$ 

$$
\frac{\pi 138^2}{207000} = 3.457 (a_c - a_0)^{-0.5}
$$
\n(2.8)

$$
a_c - a_0 = 144.56 \,\mathrm{mm} \tag{2.9}
$$

Substituting into Eq. 2.5

$$
a_c = 289.11 \,\mathrm{mm} \tag{2.10}
$$

The amount of stable crack growth (at each crack tip) that precedes failure  $(a_c - a_0)$ , Eq. 2.9

If this plate has an initial crack length  $(2a_0)$  of 50.8 mm and the plate is loaded to failure, compute the following:

Stress at failure

Using Equations. 2.5 and 2.7

$$
a_c = 2(a_c - a_0) \tag{2.11}
$$

If  $a_0 = 25.4$  mm, we get  $a_c = 50.8$  mm. Using Eq. 2.5

$$
\mathcal{G} = \frac{\pi \sigma^2 50.8}{207000} = 6.95(25.4)^{0.5}
$$
\n(2.12)

Thus, stress at failure

$$
\sigma = 213.15 \text{ MPa} \tag{2.13}
$$

The half crack size at failure  $a_c = 50.8\,\mathrm{mm}$ The stable crack growth at each crack tip  $a_c - a_0 = 25.4 \,\mathrm{mm}$